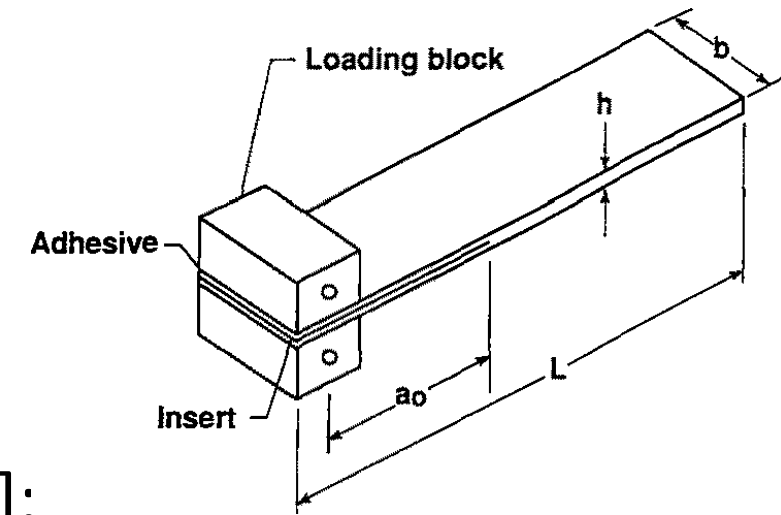


Does it really matter whether delamination toughness of polymer composites is a material parameter?

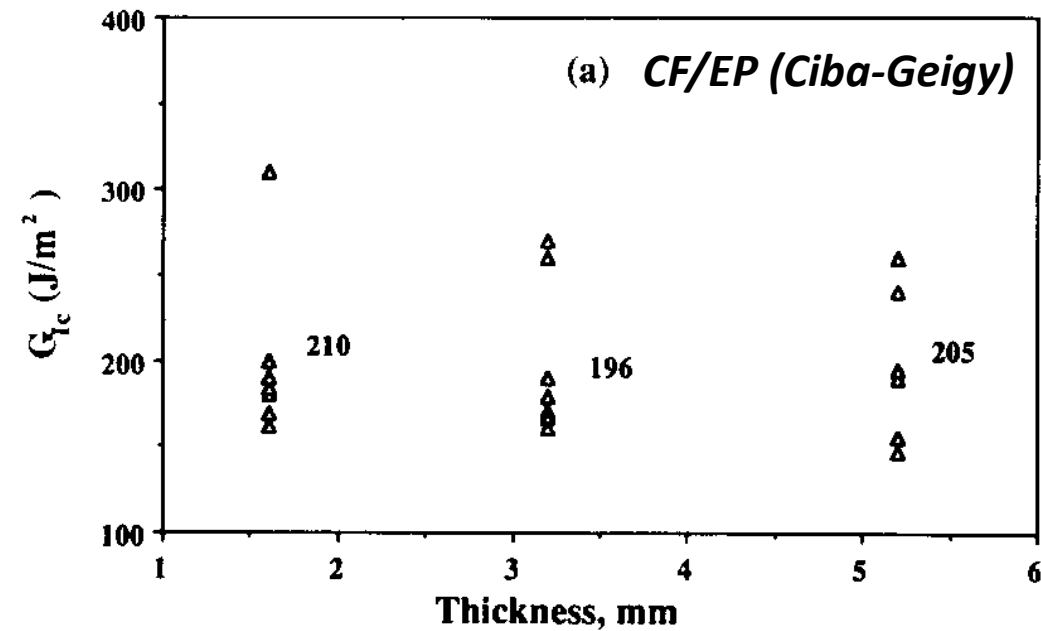
Andreas J. Brunner

*retired from Empa, Swiss Federal Laboratories
for Materials Science and Technology, Dübendorf, Switzerland*

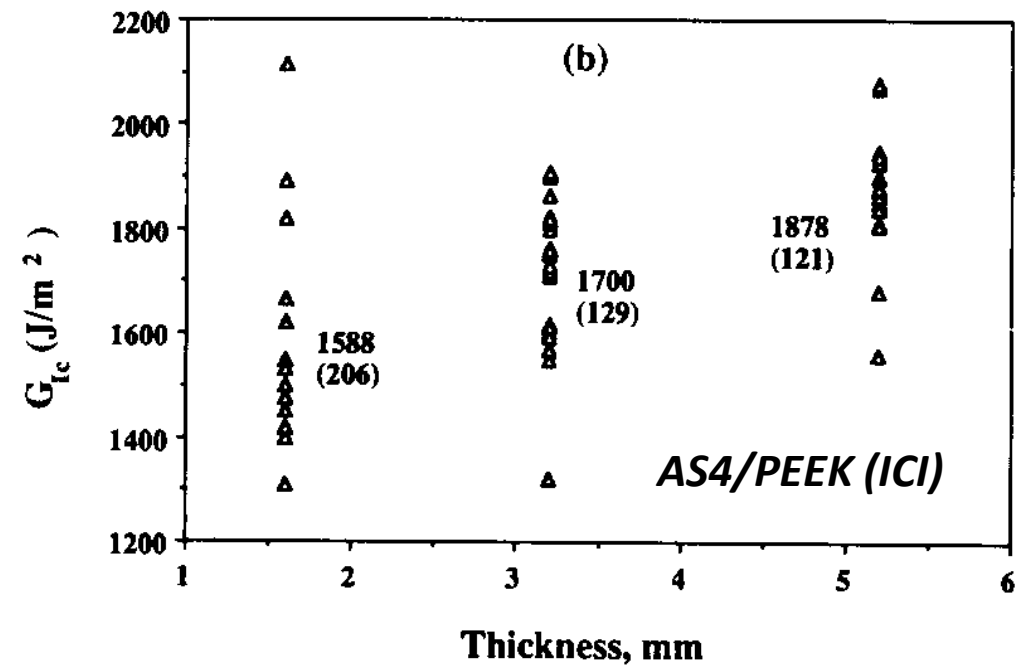
Experimental evidence I



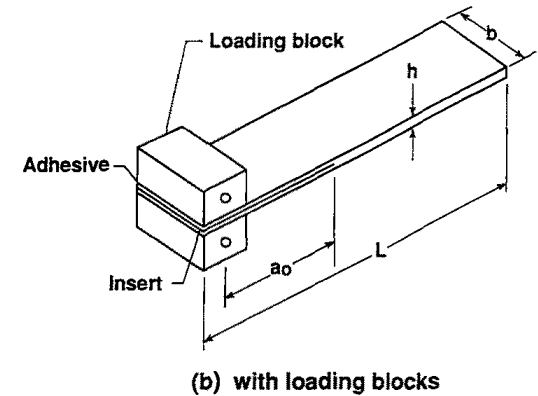
Is there a thickness effect «h»? Was investigated \approx 1990 in Mode I delamination test development [1]:



no clear thickness effect for CF/EP

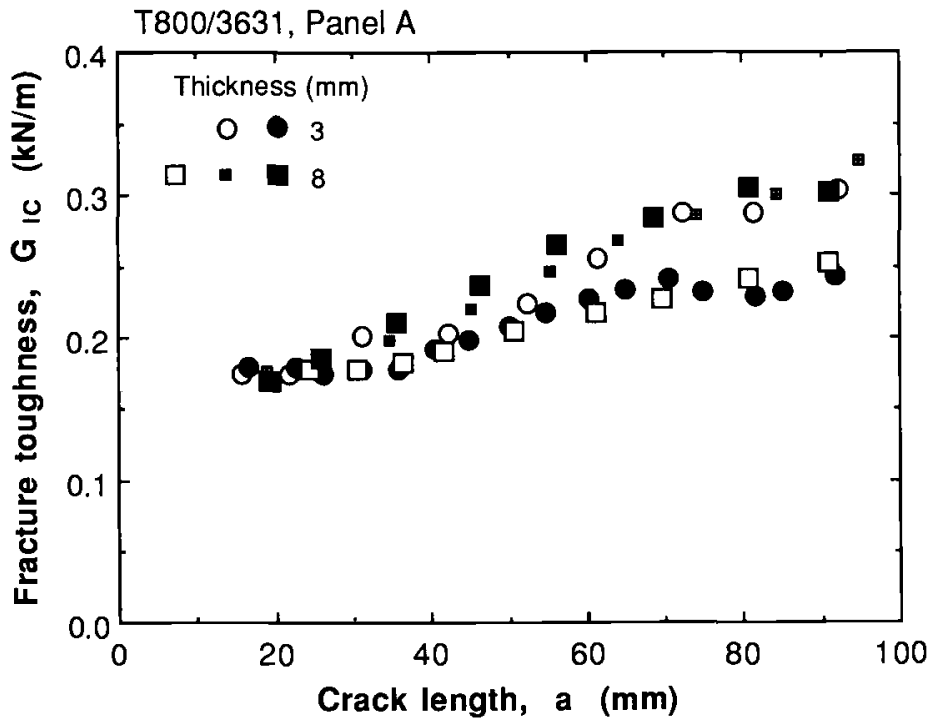


possible effect for CF/PEEK



Experimental evidence II

Are thickness effects in PEEK process-related [2] ?



8 mm laminates milled to 3 mm, 4 mm and 5 mm:
“The initiation values of the fracture toughness were independent of the specimen thickness both for AS4/PEEK laminates and T800/3631 laminates.”

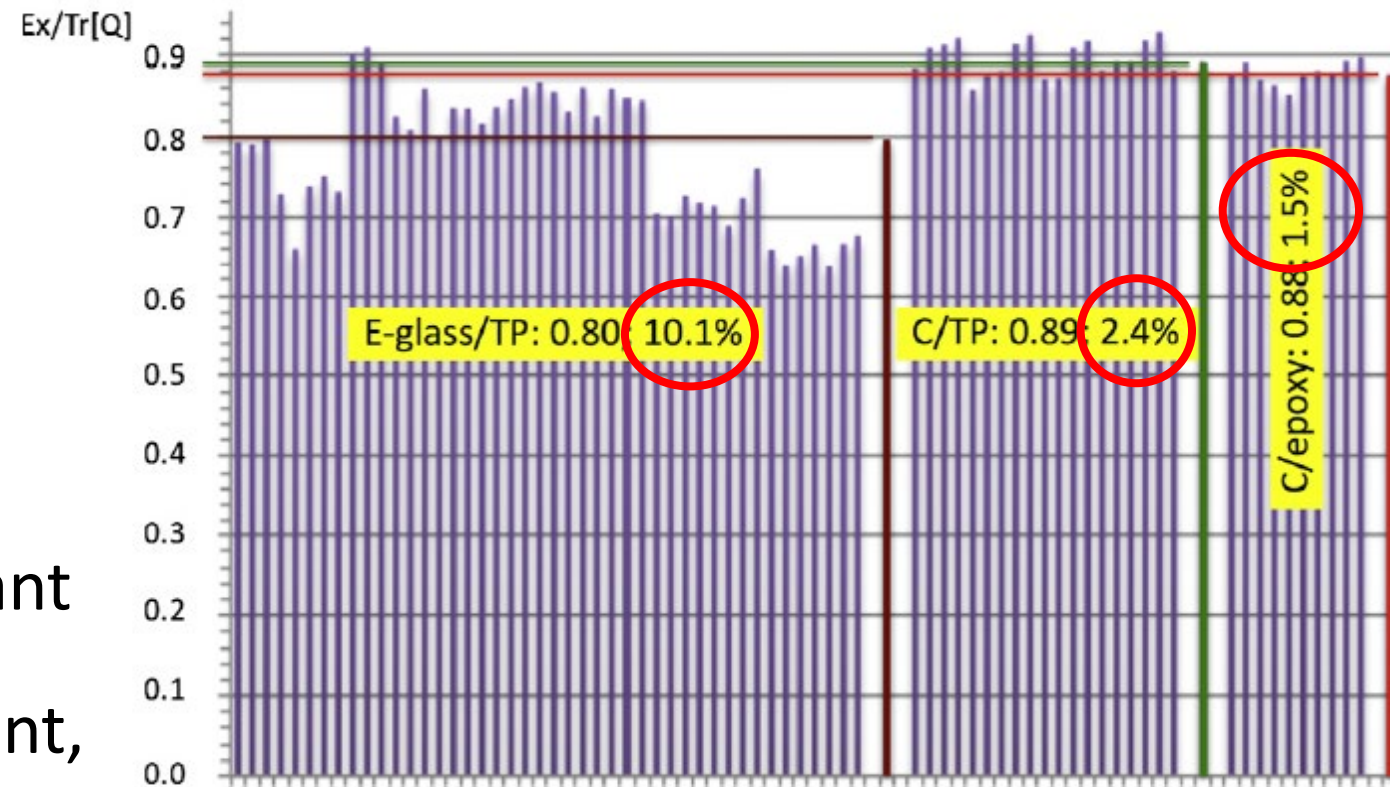
For AS4/PEEK: *“The genuine thickness effect on the propagation values obtained here was much smaller than the effect resulting from molding laminates of different thickness.”*

For CF/EP: *“The effect of specimen thickness on the propagation values at a certain crack length was smaller than the scatter of the data points for the two panels tested here.”*

[2] Hojo & Aoki 4th ASTM Symp. STP 1156 (1993) pp. 281-298

Experimental evidence III

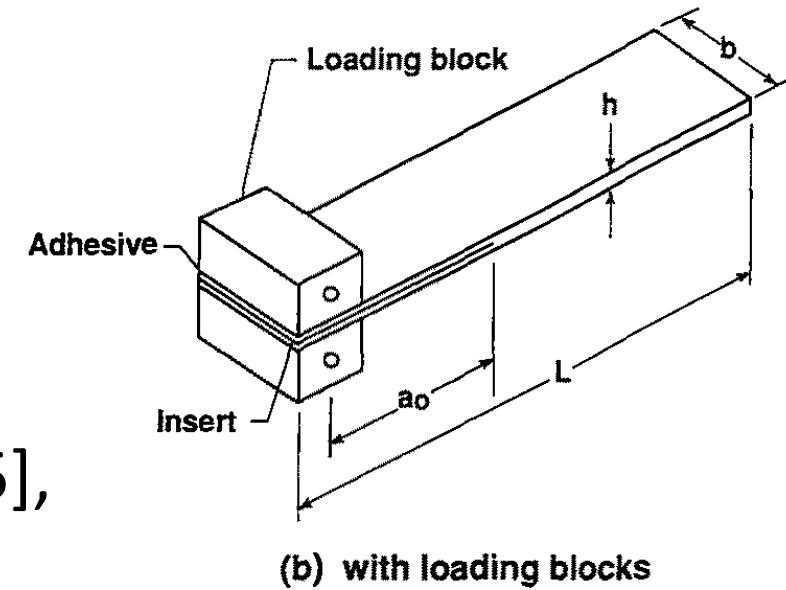
- Round robin **repeatability and reproducibility** 10-20%
- Material variability yields **intrinsic scatter**, $\approx 1-3\%$ for CFRP, $\geq 3-5\%$ for GFRP [3,4]
- **Measurement resolution** scatter estimate is $\leq 6\%$ [3]
- Additional sources of significant scatter are **human operator actions** in set-up, measurement, and data analysis [3]



[3] Brunner, Engineering Fracture Mechanics, 264 (2022) 108340

[4] Tsai and Melo, Composites Science and Technology, 100 (2014) pp. 237–243

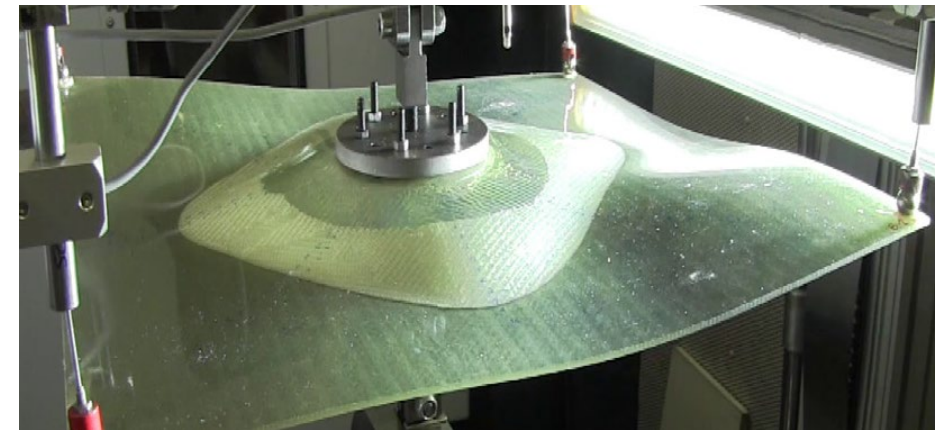
Experimental evidence IV



Is there a width effect « b »? Was investigated ≈ 1990 in Mode I delamination test development [5], widths 12.5 mm, 25.0 mm and 37.5 mm tested

*“Because **no significant width effect** was discovered 20-25 mm wide specimens were tested in the 4th and 5th rounds.”*

But Mode I 2D delamination involves membrane stresses and depends **on fiber lay-up and size and shape of loading device and precrack** [6,7]



[5] O'Brien & Martin NASA TM 104222, 1992

[6] Cameselle Molares et al. Engineering Fracture Mechanics 203 (2018) pp. 152–171

[7] Wang et al. Engineering Fracture Mechanics 250 (2021) 107787

Modelling and prediction

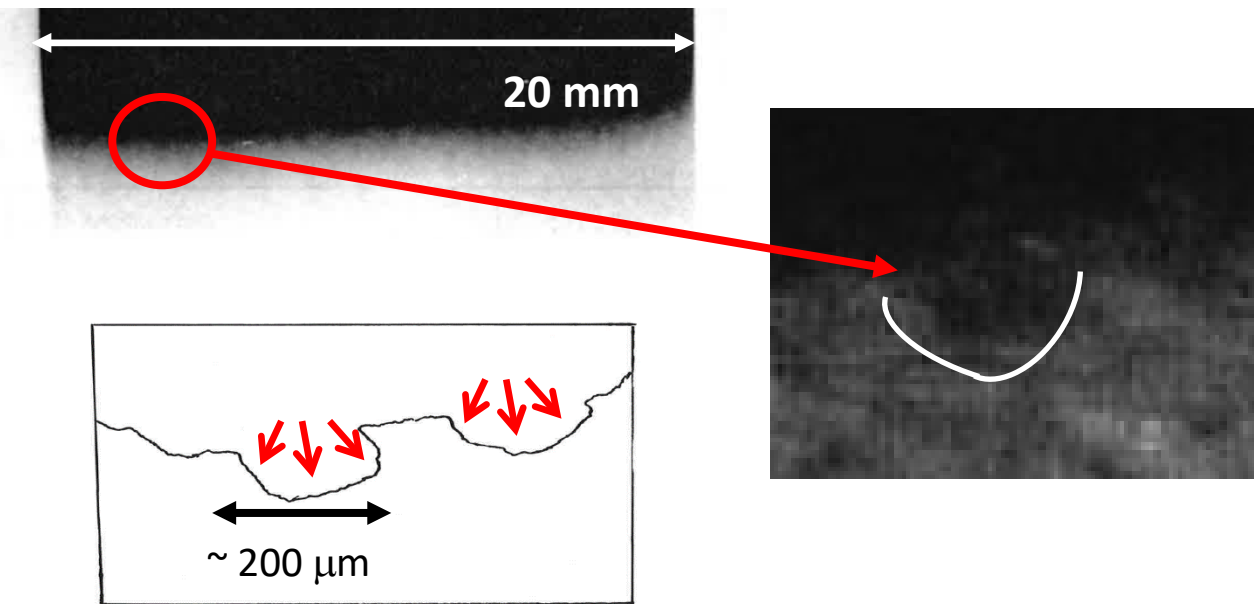
Blind modelling prediction of damage, e.g., open hole fatigue strength, yields **up to 70% scatter** depending on fiber orientation [8]

Table 2. Overall blind predictions summary of stiffness and residual strength.

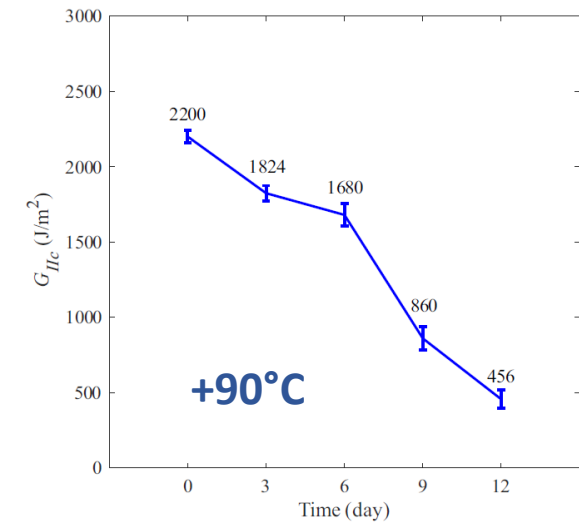
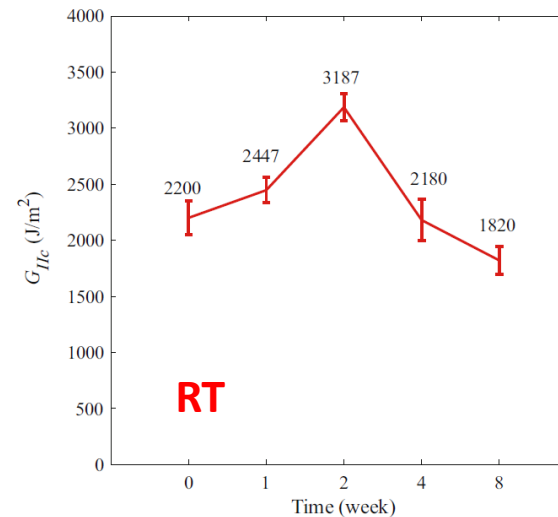
| | Experiment σ_{\max} (MPa) | GENOA σ_{\max} (MPa) | DCN σ_{\max} (MPa) | MDS-C σ_{\max} (MPa) | BSAM/MIC σ_{\max} (MPa) | MAC/GMC* σ_{\max} (MPa) | Helius PFA* σ_{\max} (MPa) | EHM* σ_{\max} (MPa) | Average error % |
|--|-------------------------------------|--------------------------------|------------------------------|--------------------------------|-----------------------------------|-----------------------------------|--------------------------------------|-------------------------------|--------------------|
| Residual strength after fatigue (200 K/300 K cycles) | | | | | | | | | |
| Open-hole tension | | | | | | | | | |
| [0/45/90/−45] _{2s} | 544 | 498 | 450 | 498 | 684 | 342 | 475 | 522 | 16 |
| [60/0/−60] _{3s} | 675 | 468 | 0 | 0 | 828 | 237 | 0 | 0 | 74 |
| [30/60/90/−60/−30] _{2s} | 474 | 384 | 383 | 0 | 408 | 303 | 428 | 307 | 26 |
| Average error (%) | | 16 | 42 | 69 | 17 | 44 | 38 | 44 | 39 |
| Open-hole compression | | | | | | | | | |
| [0/45/90/−45] _{2s} | −317 | −389 | −295 | −296 | −325 | −183 | −282 | −355 | 15 |
| [60/0/−60] _{3s} | −378 | −420 | 0 | 0 | −362 | −127 | 0 | 0 | 69 |
| [30/60/90/−60/−30] _{2s} | −274 | −380 | −245 | 0 | −332 | −165 | −249 | −248 | 33 |
| Average error (%) | | 24 | 39 | 69 | 9 | 50 | 40 | 41 | 39 |
| | Experiment E (GPa) | GENOA E (GPa) | DCN E (GPa) | MDS-C E (GPa) | BSAM/MIC E (GPa) | MAC/GMC* E (GPa) | Helius PFA* E (GPa) | EHM* E (GPa) | Average error % |

Micro-scale morphology effects

Morphology – scale estimate [9] and time-dependent behavior [10]



Average delamination damage size increment (diameter) from radiography: a **few ten to few hundred** micrometers, time-scale a **few ten** nanoseconds **to a few** microseconds



Toughness changes for constant media exposure at constant temperature: How to predict toughness for media exposure and temperature both **varying at different time-scales**?

[9] Brunner, Journal of Acoustic Emission, 33 (2016) pp. S41-S49

[10] Salamt-Talab et al. Science and Engineering of Composite Materials, 28 (2021) pp. 382–393

Conclusions

- Experimental toughness data and models both still suffer from significant scatter limiting comparison and predictions
- Multi-scale morphology, **multiple delaminations**, and time-dependent phenomena interacting on all scales yield effects on toughness observed in composite structures
- Understanding multi-scale morphology of composites and relevant interactions from **micro-/nano-scale size and time** up is essential for improving toughness prediction in composite structures from material test data
- Hence, in my opinion, it does not matter whether delamination toughness is a material parameter or not, **you have to understand** what you do when using experimental toughness data for whatever purpose

Thank you for your attention

